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# Sensor Data Qualification for Ares

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### ISHM Framework

- GRC Historical Experience in Space Propulsion HM
- Current Activities

### Sensor Data Qualification System

Description, applications to date, status and products

### Other ISHM Capabilities

- Systematic Sensor Selection Approach (S4)
- Sensors for ISHM

# Concluding Remarks

### References



# **ISHM Development Process**







# NASA Glenn Controls & Dynamics Branch System Health Management Heritage





- Automated Data Reduction / Feature Extraction SSME and Atlas/Centaur, and Post Test Diagnostics System (PTDS) for SSME and X-33
  - Significantly reduced time to analyze test data from weeks to days
- Data Quality Validation System SSME and RS-83/84
  - Demonstrated feasibility of analytical redundancy based sensor validation
- Propulsion IVHM Technology Experiment (PITEX) X34
  - Demonstrated real-time fault detection for complex propulsion system
- Propulsion Check Out and Control System (PCCS) for Integrated Propulsion Technology Demonstrator (IPTD)
- Inverse Model based Sensor Selection RS-83/84
  - Capability to optimize sensor suite for fault detection and isolation



### Current Propulsion System HM Activities in NASA Exploration Systems Programs









#### What is SDQ?

• A mathematical approach for using redundant information within a group of sensors to assess the validity of individual measurements

### Why qualify data?

 Decisions need to be made on the flight critical data in real-time and on-board the vehicle.

### What are the current approaches to onboard qualification of data?

- Only reasonableness limits are employed on all flight critical sensors
- Some redundant channel checking performed on flight control parameters
- In limited cases, driven by necessity, analytical redundancy has been applied with algorithm development tailored to a specific set of sensors.

#### What are the issues we are trying to address?

- Algorithm development in a concurrent engineering environment
- Modular flight software and generic algorithm design paradigm
- Deterministic real-time execution with static data tables that are verifiable
- Application of analytical redundancy to heterogeneous sensor set
- Need to ensure that analytical redundancy networks provide proper response to actual sensor faults AND do not incorrectly fail sensors during off-nominal system operation.





Background + Data Qualification + ISHM for J-2X + ISHM Sensors + Concluding Remarks

#### The Process

Acquire Data from sensors to be qualified and from other sources to determine system operating mode





#### **Onboard Sensor Monitoring**

Acquire Data





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#### The Process

Acquire Data from sensors to be qualified and from other sources to determine system operating mode

Estimate output of each sensor using known/derived relationships with other sensors



R1: 
$$\hat{P}_{1,1} = P_2$$
  
R2:  $\hat{P}_{1,2} = C_{2,2}P_3 + C_{2,1}$   
R3:  $\hat{P}_{2,3} = C_{3,2}P_3 + C_{3,1}$   
R4:  $\hat{W}_{1,4} = C_{4,2}(P_1 - P_3)^{1/2} + C_{4,1}$   
R5:  $\hat{W}_{1,5} = C_{5,2}(P_2 - P_3)^{1/2} + C_{5,1}$ 

#### **Onboard Sensor Monitoring**







Background + Data Qualification + ISHM for J-2X + ISHM Sensors + Concluding Remarks

### • The Process

Acquire Data from sensors to be qualified and from other sources to determine system operating mode

**Estimate** output of each sensor using known/derived relationships with other sensors **Detect** and flag breakdown of any relationships during current cycle by comparing residuals (i.e., difference between measurement & estimate) to pre-defined thresholds

if 
$$|\hat{P}_{1,1} - P_1| \le T_1$$
, then R1 = qualified, else R1 = failed  
:  
if  $|\hat{W}_{1,5} - W_1| \le T_5$ , then R5 = qualified, else R5 = failed







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#### • The Process

Acquire Data from sensors to be qualified and from other sources to determine system operating mode

Estimate output of each sensor using known/derived relationships with other sensors

**Detect** and flag breakdown of any relationships by comparing residuals (i.e., difference between measurement & estimate) to pre-defined thresholds

Decide if sensor has failed based on number and frequency of failed relationships

Bayesian Analysis-based Voting Table (Example)	No. Active ARRs for a Signal	No. Failed ARRs Required to Disqualify the Signal			
	3	3			
	4	4			
	5	4			

#### + Persistence







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### The Process

Acquire Data from sensors to be qualified and from other sources to determine system operating mode

**Estimate** output of each sensor using known/derived relationships with other sensors **Detect** and flag breakdown of any relationships by comparing residuals (i.e., difference between measurement & estimate) to pre-defined thresholds

**Decide** if sensor has failed based on number and frequency of failed relationships **Disqualify** sensor and notify system/user





# **SDQ Proof-of-Concept Design Studies**



- Demonstrated proof-of-concept for analytical redundancy-based data qualification methods using test-beds relevant to Upper Stage Subsystems.
- Demonstrated real-time hardware-in-the-loop system SDQ with deterministic execution in the presence of hardware simulated sensor and system faults
- Performed SDQ on signals with …
  - multiple sensor failures concurrent and sequential
  - bias faults on a closed-loop control feedback sensor
- Identified abnormal system operation in lieu of failing all sensors
- Characterized real-time (onboard) implementation and execution of prototype code





# **FASTR Schematic**







# **FASTR-1 SDQ ARRN\***



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ID	<b>Measurement Estimation Equations</b>
R-1	P0a = P0b
R-2	P1a = P1b
R-3	P1a = <i>a</i> + <i>b</i> *P2
R-4	P1b = a + b*P2
R-5	OR1 = <i>a</i> + sign(P0a-P1a)* <i>b</i> *( P0a-P1a ) <sup>1/2</sup>
R-6	OR1 = $a + sign(P0a-P1b)*b*( P0a-P1b )^{1/2}$
R-7	OR1 = $a + sign(P0a-P2)*b*( P0a-P2 )^{1/2}$
R-8	OR1 = $a + sign(P0b-P1a)*b*( P0b-P1a )^{1/2}$
R-9	OR1 = $a + sign(P0b-P1b)*b*( P0b-P1b )^{1/2}$
R-10	OR1 = $a + sign(P0b-P2)*b*( P0b-P2 )^{1/2}$
R-11	CBP = a + b*P0a
R-12	$CBP = a + b^*P0b$

\*Analytical Redundancy Relationship Network











# SDQ Results for Test Series 2 P1a & P1b Common Cause Failure



Test	Fault	FASTR	-1	FASTR-2		FASTR-3	
Number	Description	SDQ Output	Cycles	SDQ Output	Cycles	SDQ Output	Cycles
TS2_01		P1a P1b OR1 Suspended	70	P1a P1b OR1 <b>Abnormal</b>		P1a P1b Suspended	70
TS2_02	Blocked Sensing Port	OR1	18	P1a P1b OR1 Suspended	18	P1a P1b Suspended	18
TS2_03	Positive Level Shift	<b>OR1</b> P1a	5	OR1 P1b P1a <b>Abnormal</b>		P1a P1b Suspended	6
TS2_04		OR1	13	P1a P1b OR1 Suspended	16	P1a P1b Suspended	17



# Comparison of Detection Statistics for FASTR ARRNs



Test	No.	FASTR-1			FASTR-2			FASTR-3		
Series	Tests	Green	Yellow	Red	Green	Yellow	Red	Green	Yellow	Red
1	13	100%	0%	0%	62%	38%	0%	92%	0%	8%
2	4	0%	25%	75%	0%	25%	75%	100%	0%	0%
3	2	100%	0%	0%	100%	0%	0%	100%	0%	0%
4	2	0%	0%	100%	0%	0%	100%	<b>50%</b>	0%	<b>50%</b>
5	8	50%	50%	0%	50%	13%	38%	75%	<b>25%</b>	0%
Combined	29	66%	17%	17%	48%	24%	28%	86%	7%	7%





- Investigated Alternate Analytical Redundancy Relationship Networks (ARRNs) for Qualification of Flow and Pressure Sensors
  - Qualified heterogeneous sensors in ARRN consisting of five (5) pressure measurements and one (1) flow measurement
  - Trained ARRNs using test data with "real-world" measurement variations
  - Tested detection capabilities using hardware simulated faults
  - Demonstrated capability/limitations in detecting multiple concurrent and sequential sensor failures.

# Lessons Learned

- Use of significantly less certain measurements to qualify more certain measurements can degrade qualification accuracy.
- Use of ARRs that are not truly independent can result in an ARRN that is prone to incorrect detections & network suspensions.
- Lack of sufficient relationships can result in missed detections.





# Systematic Sensor Selection Strategy (S4)

- Quantitative approach for reducing sensors to the "minimal necessary set"
- Proactive means of circumventing the costly addition of sensors after the system design has been finalized
- Efficiently searches large, complex solution spaces
- Incorporates critical FMEA and risk information

# **S4** Applications

- PWR J2X engine for Ares Upper Stage
- Rocketdyne RS-83 and RS-84 engines
- Advanced sensors for turbine engines







# **Objective:**

 Develop S4 detectability database for critical faults using large set of candidate J-2X sensors

# **GRC Contribution**

- Added new capabilities to PWR J-2X Real-Time Model
  - Simulation of user-defined (e.g., off-nominal) operating mode profiles
  - Simulation of user-defined faults (bias and drift, multiple sequential or concurrent)
  - Variation of propellant inlet conditions
- Developed Excel-based User Interface
- Provided independent verification of results

# Benefit to J-2X Program

- Allows J-2X program to successfully meet milestones
  - Enhances overall simulation capability and reduces time to generate simulation data
  - Provides mechanism for investigating propagation and detectability of complex faults
  - Generates results that are more "flight-like"
  - Automates simulation setup, execution, and analysis
  - Ensures simulation results were generated correctly



# Systematic Sensor Selection Strategy for J-2X Test Data Reduction



Background + Data Qualification + ISHM for J-2X + ISHM Sensors + Concluding Remarks



#### Systematic Sensor Selection Strategy is used to analytically verify that J-2X test sensor suite will meet test data reduction goals

- Accurately characterize J-2X engine operation over multiple operating states
- Calibrate each engine to meet thrust & mixture ratio targets
- Enable detection of off-nominal engine component operation which might result in faulty calibration
- Support engine model calibration for specific engine configurations
- Ensure that acceptance of production J-2X engine systems is based on reliable test data
   J-2X En





#### **Data Reduction Process**



#### HARSH ENVIRONMENT ELECTRONICS AND SENSORS APPLICATIONS UNIQUE RANGE OF HARSH ENVIRONMENT TECHNOLOGY AND CAPABILITIES Background & Data Qualification & ISHM for J-2X & ISHM Sensors & Concluding Remarks



- OPERATION IN HARSH ENVIRONMENTS
- ALL-IN-ONE SHOP FOR HARSH ENVIRONMENT SYSTEM APPLICATIONS
- RANGE OF PHYSICAL AND CHEMICAL MEASUREMENTS
- HARSH ENVIRONMENT MICROSYSTEMS TECHNOLOGY
- INCREASE DURABILITY, DECREASE THERMAL SHIELDING, IMPROVE IN-SITU OPERATION
- ENABLE EXPANDED MISSION PARAMETERS

#### Range of Physical and Chemical Sensors for Harsh Environments and Intelligent Systems



High Temp Fiber Sensor Operation



High Temperature 7 Pressure Sensor 8 National Aeronautics and Space Administration



Self Diagnostic Accelerometer



Thin Film Sensors

#### High Temperature Electronics and Wireless





World Record High Temperature Electronics

High Temperature RF Components

Energy Harvesting Thin Film Thermoelectrics

#### World Recognized Accomplishments





1998 R&D 100 Award 2004 R&D 100 Award





1995 R&D 100 Award 1991 R&D 100 Award

Long Term Vision: High Temperature "Lick and Stick" Systems







### Hydrogen Sensor Technology

- MICROFABRICATED USING MEMS-BASED TECHNOLOGY FOR MINIMAL SIZE, WEIGHT AND POWER
- HIGHLY SENSITIVE IN INERT/OXYGEN ENVIRONMENTS, WIDE CONCENTRATION RANGE DETECTION

MATURE TECHNOLOGY IMPLEMENTED, E.G., ON THE ISS



# "Lick And Stick" Leak Sensor System

• CONFIGURATIONS INCLUDE THREE SENSORS, SIGNAL CONDITIONING,

POWER, AND TELEMETRY IN A SINGLE PACKAGE

H<sub>2</sub> COTS SYSTEM SCHEDULED FOR CLV IMPLEMENTATION

Micro-Fabricated Gas Sensors for Low False Alarm Fire Detection

NASA 2005 TURNING GOALS INTO REALITY AA'S CHOICE 2005 R&D 100 AWARD WINNER

 MEMS-BASED CHEMICAL SPECIES AND PARTICULATE SENSORS ORTHOGONAL DETECTION SIGNIFICANTLY REDUCES FALSE ALARMS CEV FIRE AND ENVIRONMENTAL MONITORING



Sensor

MEI Makel Engineering Inc.

Sensor

Sensor







### NASA Glenn is developing key technologies for access to space propulsion system ISHM

- Sensor Data Qualification (SDQ) to ensure that critical flight operations are based on validated data
- Systematic Sensor Selection Strategy (S4) to determine "optimal" sensor suite for Fault Detection and Isolation
- Harsh environment electronics/sensors and chemical sensors support system health assessment.

# SDQ, S4 and sensor technologies being applied to NASA Exploration System programs

- SDQ approved by Ares Upper Stage Chief Engineer for onboard qualification of flight critical sensors.
- S4 used to analytically verify that J-2X test sensor suite will meet test data reduction and engine calibration goals
- H<sub>2</sub> sensor (mature technology) demonstrated on ISS, COTS version scheduled for Ares implementation

# NASA interested in further technology maturation and application opportunities





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